This Page Is Inserted by IFW Operations and is not a part of the Official Record

BEST AVAILABLE IMAGES

Defective images within this document are accurate representations of the original documents submitted by the applicant.

Defects in the images may include (but are not limited to):

- BLACK BORDERS
- TEXT CUT OFF AT TOP, BOTTOM OR SIDES
- FADED TEXT
- ILLEGIBLE TEXT
- SKEWED/SLANTED IMAGES
- COLORED PHOTOS
- BLACK OR VERY BLACK AND WHITE DARK PHOTOS
- GRAY SCALE DOCUMENTS

IMAGES ARE BEST AVAILABLE COPY.

As rescanning documents will not correct images, please do not report the images to the Image Problems Mailbox.

THIS PAGE BLANK (USPTO)

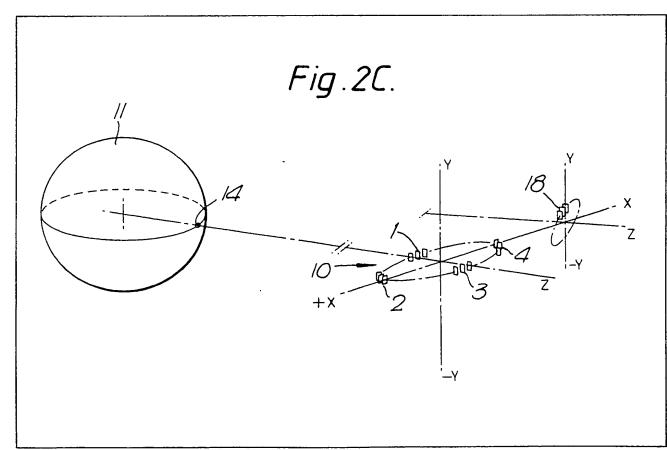
UK Patent Application (19) GB (11) 2 134 353 A

- (21) Application No 8400768
- (22) Date of filing 12 Jan 1984
- (30) Priority data
- (31) 8300747
- (32) 12 Jan 1983
- (33) United Kingdom (GB)
- (43) Application published 8 Aug 1984
- (51) INT CL3
 - H04B 7/185
- (52) Domestic classification H4L RR U1S 2212 H4L
- (56) Documents cited None
- (58) Field of search
 - H4L
- (71) Applicant
 British Aerospace Public
 Limited Company,
 (United Kingdom),
 100 Pall Mall,
 London SW1Y 5HR.
- (72) Inventor Ronald William Young

- (74) Agent and/or Address for Service D. J. Saul,
 - British Aerospace PLC, Corporate Patents
 - Department,
 Brooklands Road,
 - Weybridge,
 - Surrey, KT13 0SJ.

(54) Co-operative satellites

(57) A satellite orbiting arrangement including an orbiting cluster 10 of cooperative satellites in which intersatellite communication is to be maintained substantially constantly, the arrangement further including an intersatellite communication satellite 18 positioned remotely from the cluster in an orbit selected to maintain at least approximate station with the cluster and to enable communication with each satellite of the cluster to be maintained substantially constantly.



The drawing(s) originally filed was (were) informal and the print here reproduced is taken from a later filed formal copy.

GB 2 134 353 A

Fig.1.

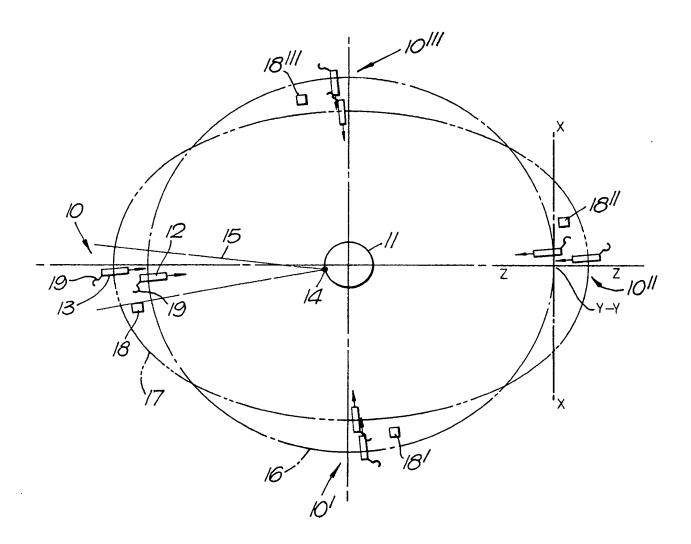


Fig.2A.

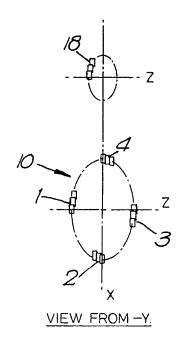
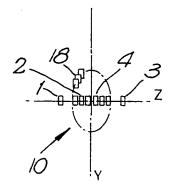


Fig.2B.



VIEW FROM +X.

Fig. 2C.

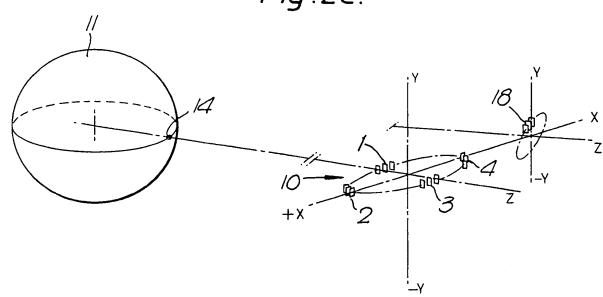
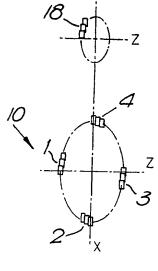
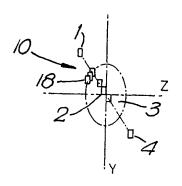


Fig. 3A.



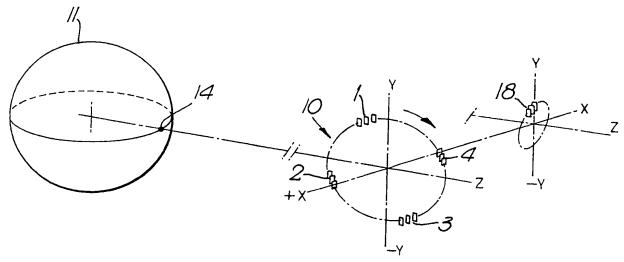
VIEW FROM -Y.

Fig.3B.



VIEW FROM +X.

Fig.3C.



SPECIFICATION

Co-operative satellites

5 This invention relates to co-operative satellites.
 Co-operative satellites are an alternative to very large satellites which require very large launch vehicles not presently available at least commercially, and to very large satellite structures (sometimes
 10 known as space stations), which can be assembled in space from small, more easily launched components.

Such co-operative satellites are arranged, when in use to orbit, as a cluster of individual satellites

15 located in such proximity that in general they cannot be individually resolved by a communications ground station. In order that the individual satellites remain co-located certain criteria must be met, bearing in mind that each satellite in the cluster

20 circles the Earth on its own individual orbit as if the other cluster satellites were absent. The criteria are

- (a) the semi-major axes of each individual orbit must be equal, and,
- (b) the mean longitudes of each satellite must be 25 approximately equal.

Thus, the individual satellites are co-located and the orbit periods of each are equal so that the cluster does not disperse with time.

Since the individual satellites in effect replace a
30 single unit (i.e. a large satellite or space station), the
otherwise single payload will be distributed
amongst them. This necessitates an inter-satellite
communications network between the members of
the cluster; this is conveniently achieved by desig35 nating one satellite of the cluster as an inter-satellite
communications exchange or network hub through
which the inter-satellite communications can be
routed. Naturally, one criteria is that the intersatellite communication satellite shall be always in
40 clear line of sight with each of the individual
satellites of the cluster so that inter-satellite communications are maintained without interruption.

The individual satellites of a cluster follow their individual orbits, which although similar, are not 45 identical and, futhermore, because each satellite is directed towards the earth-based communications station at all times, it is found that the line of sight between any two cluster members rotates through a total of 360° during an orbit period. This is for 50 satellites having the same mean longitude. This, in turn, requires that in order to maintain continuous communication between the individual satellites and the inter-satellite communication satellite, each satellite of the cluster must carry a receptor (e.g. an 55 antenna) capable of 360° scanning movement. This requirement is most disadvantageous from the engineering aspect in a satellite which has tightly packed equipment internally and many essential protrusions externally.

According to one aspect of the present invention, a satellite orbiting arrangement includes an orbiting cluster of co-operative satellites in which intersatellite communication is to be maintained substantially constantly, the arrangement further including an inter-satellite communication satellite positioned

remotely from the cluster in an orbit selected to maintain at least approximate station with the cluster and to enable communication with each satellite of the cluster to be maintained substantially constantly.

According to a further aspect of the invention, satellite apparatus for placing in orbit includes a plurality of co-operative satellites which, when in use, orbit as a cluster and require to have intersatellite communication maintained substantially constantly; and further including an inter-satellite communication satellite which, when in use, is positioned remotely from the cluster in an orbit selected to maintain at least approximate station with the cluster and to enable communication with each satellite of the cluster to be maintained substantially constantly.

Some embodiments of the invention are now described by way of example with reference to the accompanying drawings in which:

Figure 1 illustrates, diagrammatically and not to scale, a view on the North Pole of the Earth looking towards the South Pole with a cluster of individual satellites in orbit within an equatorial plane; the 90 cluster is shown four times in varying positions,

Figures 2A, 2B and 2C are different views of satellite movement in an earth-fixed frame when the satellites are in a cluster of equatorial ellipse geometry, that is to say they lie always in an 95 equatorial plane and follow one another around a planar elliptical path in the earth-fixed frame.

Figures 3A, 3B and 3C are similar views to those of Figurtes 2A, 2B and 2C but showing the satellites in a cluster of tilted circule geometry, that is to say they deviate from the equatorial plane and follow one another around a planar circular path in the earth-fixed frame.

Referring initially to Figure 1, a cluster 10 of satellites are shown in nominal geostationary orbit around the earth 11. For this, the radius of orbit of the cluster is set at approximately 42,000 kilometres in an equatorial plane so that the cluster thus has a period of 24 hours, the same as that of the Earth; the cluster thus appears to be substantially stationary above one spot on the Earth's equator within sight of which a communication station 14 is sited. The cone 15 of "illumination" of the communication station includes all the satellites of the cluster.

For simplicity, this Figure shows only two satellites 12, 13 in the cluster. That referenced 12 is shown by way of example in a perfect circular orbit 16 whilst that referenced 13 is shown slightly displaced from that perfect orbit as would be necessary in any cluster geometry; it therefore follows an elliptical orbit 17. It will be recalled that in a cluster each satellite follows its own orbit, that precise orbit being dependent upon the actual position of the satellite with respect to the truly geostationary orbit.

The cluster is shown in four different positions 10, 10', 10", 10"; from these it can be seen that if the satellites are directed always at the earth station 14, as is necessary for them to provide their international communications function, any inter-satellite communication network will require antenna dishes 19

which move angularly through a total of 360° with respect to the satellite body on which it is mounted during each orbital period to give a clear line of sight at all times. As before discussed, this 360° scan is disadvantageous from the engineering aspect; it is difficult to provide.

This relative movement with respect to one another is further illustrated in Figures 2 and 3.

In these Figures, +Z points towards the Earth's

10 centre, +Y points South normal to the Earth's
equatorial plane, and +X points East along the
geostationary orbit (i. e. tangential thereto). In all
cases, the frame of reference is earth-fixed. The XYZ
co-ordinates are also marked on Figure 1 for ease of
15 reference.

In these Figures, four satellites 1, 2, 3 and 4 are shown in the cluster. They are similar to those referenced 12 and 13. Each satellite 1, 2, 3, 4 is illustrated three times at 0, 0.02 and 0.04P where P is the orbital period of 24 hours.

The Figures 2 show an equatorial ellipse geometry in which the relative movement of members of the cluster in the earth-fixed frame is elliptical, the plane of the ellipse lying in the earth's equatorial plane as is particularly shown in Figure 2B. These Figures 2 show the relative movements of satellites in the

orbits similar to those of Figure 1.

The Figures 3 show a tilted circle geometry (an alternative could be a tilted ellipse, for example), in 30 which the satellites are not only displaced in the equatorial plane, but in inclination as well, the plane of the circule (or ellipse) thus lies tilted to the equatorial plane as is particularly shown in Figure 3B.

35 Irrespective of the geometry of the cluster it is convenient to designate one satellite being part of or in station with the cluster for inter-satellite communications network purposes. This will be an exchange or network hub satellite having a switch-

40 ing and relay function, so that the individual satellites of a cluster can route their inter-satellite communications through it. This arrangement still requires inconveniently large scanning movements of receptors/antennas. To reduce this movement, the
 45 inter-satellite communication satellite is positioned

5 Inter-satellite communication satellite is positioned remotely from the cluster.

This is illustrated in Figure 1, where such a satellite is diagrammatically shown in four positions 18, 18', 18", and 18". Even though its orbit is not identical 50 with all (or necessarily any) members of the cluster, it is evident that the scanning angles of the intersatellite communication antennas 19 are reduced to angles much more conducive to an engineering solution.

55 The satellite 18 may or may not be within the ground station cone of "illumination" 15.

The satellite 18 may be positioned in a geostationary orbit similar to that shown for satellite 12, depending upon the orbits chosen for other mem60 bers of the cluster and bearing in mind the requirement for a clear sight line at all times. It may, alternatively, be placed in such an orbit that it follows an equatorial circular or elliptical geometry as evidenced by those satellites 1, 2, 3 and 4 of
65 Figures 2 or a tilted circular or elliptical geometry as

evidenced by those satellites 1, 2, 3, and 4 of Figures 3. Any of these options are selectable to provide a clear sight line between the satellite 18 and the members of the cluster 10, whilst maintaining a 70 relatively small scan angle for the antennas 19.

In this respect, it may be advantageous if the plane of the ellipse or circule of the relative motion of the satellite 18 is inclined to that of the cluster 10. An example of such an arrangement is shown superimposed on Figures 2B and 3B, as satellite 18 with an elliptical relative path when viewed along axis X and to the right of Figures 2C and 3C.

It is possible to include a spare inter-satellite communication satellite 18 in close promitity orbit to original satellite 18. Moreover, the satellite 18 could well provide inter-satellite communications for a further cluster (not shown) additional to that referenced 10, or for providing communications to other clusters, satellites or space stations.

CLAIMS

85

- A satellite orbiting arrangement including an orbiting cluster of co-operative satellites in which inter-satellite communication is to be maintained substantially constantly, the arrangement further including an inter-satellite communication satellite positioned remotely from the cluster in an orbit selected to maintain at least approximate station with the cluster and to enable communication with each satellite of the cluster to be maintained substantially constantly.
- A satellite orbiting arrangement according to Claim 1, wherein both the cluster and the communication satellite are in substantially geo-stationary orbit.
- A satellite orbiting arrangement according to Claim 2, in which the orbits are selected so that individual satellites of the cluster follow each other around a path lying generally in a single plane with reference to an earth fixed frame.
- 4. A satellite orbiting arrangement according to Claim 3, in which the orbit of the communication satellite is selected so that it follows a path lying
 110 generally in a plane which, with reference to the earth-fixed frame, is different to that of the cluster.
- 5. Satellite apparatus for placing in orbit including a plurality of co-operative satellites which, when in use, orbit as a cluster and require to have
 115 inter-satellite communication maintained substantially constantly, and further including an intersatellite communication satellite which, when in use, is positioned remotely from the cluster in an orbit selected to maintain at least approximate station
 120 with the cluster and to enable communication with each satellite of the cluster to be maintained substantially constantly.

Printed for Her Majesty's Stationery Office, by Croydon Printing Company Limited, Croydon, Surrey, 1984. Published by The Patent Office, 25 Southampton Buildings, London, WC2A 1AY, from which copies may be obtained.